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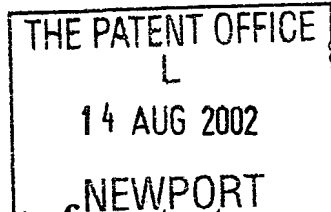
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Dated 30 June 2003



14AUG02 E740821-1 D00370
P01/7700 0.00-0218849.8

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The Patent Office

Cardiff Road
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South Wales
NP10 8QQ

1. Your reference

DY3016

2. Patent application number

(The Patent Office will fill in this part)

0218849.8

3. Full name, address and postcode of the or of each applicant (underline all surnames)

ROLLS-ROYCE PLC
65 BUCKINGHAM GATE
LONDON
SW1E 6AT

Patents ADP number (if you know it)

ENGLAND

If the applicant is a corporate body, give the country/state of its incorporation

4. Title of the invention

LUBRICATION SYSTEM FOR GAS TURBINE ENGINE

5. Name of your agent (if you have one)

M A GUNN

"Address for service" in the United Kingdom to which all correspondence should be sent (including the postcode)

ROLLS-ROYCE plc
PATENTS DEPARTMENT
PO BOX 31
DERBY
DE24 8BJ

Patents ADP number (if you know it)

6. If you are declaring priority from one or more earlier patent applications, give the country and the date of filing of the or of each of these earlier applications and (if you know it) the or each application number

Country

Priority application number
(if you know it)

Date of filing
(day / month / year)

7. If this application is divided or otherwise derived from an earlier UK application, give the number and the filing date of the earlier application

Number of earlier application

Date of filing
(day / month / year)

8. Is a statement of inventorship and of right to grant of a patent required in support of this request? (Answer 'Yes' if:

YES

a) any applicant named in part 3 is not an inventor, or

b) there is an inventor who is not named as an applicant, or

c) any named applicant is a corporate body.

See note (d))

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Continuation sheets of this form

Description	12
Claim(s)	3
Abstract	1
Drawing(s)	6

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Priority documents

Translations of priority documents

Statement of inventorship and right to grant of a patent (Patents Form 7/77) 3

Request for preliminary examination and search (Patents Form 9/77) 1

Request for substantive examination (Patents Form 10/77) 1

Any other documents (please specify)

DEPOSIT ACCOUNT FEE SHEET

11.

I/We request the grant of a patent on the basis of this application.

Signature

M A GUNN

Date

13.8.2002

12. Name and daytime telephone number of person to contact in the United Kingdom

P D GEE

01332 249638

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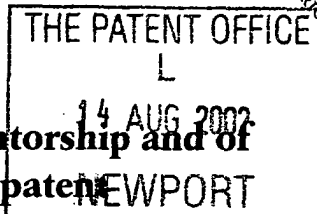
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**Statement of inventorship and of
right to grant of a patent**



The Patent Office

Cardiff Road
Newport
South Wales
NP10 8QQ

1. Your reference

DY3016

2. Patent application number
(if you know it)

0218849.8

14 AUG 2002

3. Full name of the or of each applicant

ROLLS-ROYCE PLC

4. Title of the invention

LUBRICATION SYSTEM FOR GAS TURBINE ENGINE

5. State how the applicant(s) derived the right
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BY VIRTUE OF AN AGREEMENT DATED 11 DECEMBER 2000

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2

7.

I/We believe that the person(s) named over the page (and on
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Signature

Date

M A GUNN

13.8.2002

8. Name and daytime telephone number of
person to contact in the United Kingdom

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Patents ADP number (if you know it): 760985 2001

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A Lubrication System

This invention relates to lubrication systems. More particularly, but not exclusively, this invention relates to a method and apparatus for controlling the flow of lubricating oil in a gas turbine engine.

Conventionally, gas turbine engine oil systems have a continuous oil flow which feeds to bearings, gears and splines. Also, components of a gas turbine engine that are manufactured from non-corrosion resistant materials are provided with oil to prevent corrosion. Most gas turbine engines use a self-contained recirculatory lubrication system in which the oil is distributed around the engine and returned to an oil tank by pumps. However, some engines use what is known as a total loss or expendable system in which the oil, after lubricating the engine, is either burnt in the engine's normal combustion process or is allowed to spill overboard.

In known arrangements, oil is supplied to desired locations on the basis of that needed during take-off conditions where the performance of the engine is at its maximum. However, a gas turbine engine operates for a large proportion of its cycle at cruise conditions, which requires less performance and hence a lower supply of oil to its components than at take-off.

It is known to have a lubrication system of this type where the maximum oil flow is not supplied to the engine components all the time. In JP 58051299, for example, a system is described in which a solenoid valve permits a proportion of the oil flow from the pump to be recirculated to the pump inlet, when the engine is running at low power. Recirculating systems such as this have the disadvantage that the pump is operating at full power all the time, which increases power consumption, noise and oil heating.

US 4245465 discloses a three-stage poppet valve to modulate the oil flow in response to engine speed. US 5180034 discloses a system with a variable-speed oil pump, controlled in response to engine speed, throttle setting
5 and other parameters, to vary the oil flow to the engine components.

Systems that modulate the flow of oil in this way have the disadvantage that, where oil is directed to a specific region within a component (for example, within a bearing
10 chamber) by means of a targetted jet or nozzle, the jet of oil will only be aimed correctly for one flow condition, and when the flow is modulated by the control system to a different condition, the aim of the oil jet will be upset and the component will not be supplied with oil in the
15 optimum manner.

According to one aspect of the present invention, a lubrication system comprises a reservoir for lubricant, lubricant delivery means to deliver lubricant to at least one component to be lubricated, an injector associated with
20 each component, a control system to control the opening and closing of each injector and scavenge means to remove surplus lubricant from around each component and return it to the reservoir, in which each injector comprises a pulsative injector, controlled by the control system to
25 deliver lubricant to its associated component in a series of discrete pulses.

At least one injector may be a multiple step injector. At least one injector may be a variable aperture injector. At least one injector may have a minimum flow bypass.

30 In a preferred embodiment of this aspect of the invention, the control system controls the supply of lubricant to at least one component in response to the operating conditions of that component.

The lubricant delivery means may comprise a pump.

The lubricant delivery means may comprise at least one accumulator to smooth the fluctuations in pressure arising out of the pulsative delivery of lubricant.

The scavenge means may comprise a variable speed
5 scavenge pump maintaining a substantially constant pressure head and a restrictor associated with each component.

At least one restrictor may be a variable restrictor controlled by the control system in response to the operating conditions of its associated component.

10 The scavenge means may comprise a scavenge pump associated with each component, each scavenge pump being controlled by the control system in response to the operating conditions of its associated component.

At least one scavenge pump may be a constant volume
15 pump. At least one scavenge pump may be a constant speed, variable volume pump.

The lubrication system may be a lubrication system for a gas turbine engine.

According to one embodiment of an alternative aspect
20 of the invention, a method for lubricating a component of a gas turbine engine comprises the steps of storing lubricant in a reservoir, delivering the lubricant to a pulsative injector associated with the component, controlling the pulsative injector to deliver lubricant to the component in
25 a series of discrete pulses and scavenging surplus lubricant from the component and returning it to the reservoir.

According to an alternative embodiment of this aspect of the invention, a method for lubricating a component of a
30 gas turbine engine comprises the steps of storing lubricant in a reservoir, delivering the lubricant to a pulsative injector associated with the component, sensing at least one first condition of the component, controlling the pulsative injector, in response to at least one first

condition of the component, to deliver lubricant to the component in a series of discrete pulses and scavenging surplus lubricant from the component and returning it to the reservoir.

5 Preferably the scavenging of surplus lubricant comprises the steps of sensing at least one second condition of the component and controlling a scavenge pump associated with the component, in response to at least one second condition of the component, to scavenge surplus
10 lubricant from the component and return it to the reservoir. The second condition may be one or more selected from the group comprising a pressure, a temperature, a heat flux, an engine speed.

According to a further alternative embodiment of this
15 aspect of the invention, a method of lubricating a component of a gas turbine engine comprises lubricating the component by a lubricating system according to any of the seventh to sixteenth paragraphs above.

The invention will now be more particularly described,
20 by way of example, with reference to the accompanying drawings in which:

Figure 1 is a cross section through a gas turbine engine of known type;

Figure 2 is a cross sectional view through part of a
25 gas turbine engine showing part of a lubrication system according to the present invention;

Figure 3 is a cross sectional view through part of a gas turbine engine showing part of an alternative embodiment of a lubrication system according to the present
30 invention;

Figure 4 is a schematic illustration of a lubrication system of known type;

Figure 5 is a schematic illustration of one embodiment of a lubrication system according to the invention;

Figure 6 illustrates schematically a number of possible schedules for the opening and closing of an injector in a lubrication system according to the invention; and

5 Figure 7 is a schematic illustration of an alternative embodiment of a lubrication system according to the invention.

Referring first to Figure 1, a gas turbine engine shown generally at 10 is of conventional configuration. It
10 comprises in axial flow series a fan 11, intermediate pressure compressor 12, high pressure compressor 13, combustor 14, high, intermediate and low pressure turbines 15, 16 and 17 respectively and an exhaust nozzle 18.

Air is accelerated by the fan 11 to produce two flows
15 of air, the outer of which is exhausted from the engine 10 to provide propulsive thrust. The inner flow of air is directed into the intermediate pressure compressor 12 where it is compressed and then directed into the high pressure compressor 13 where further compression takes place.

20 The compressed air is then mixed with fuel in the combustor 14 and the mixture combusted. The resultant combustion products then expand through the high, intermediate and low pressure turbines 15, 16, 17 respectively before being exhausted to atmosphere through
25 the exhaust nozzle 18 to provide additional propulsive thrust. The high, intermediate and low pressure turbines 15, 16, 17 drive, respectively, the high and intermediate pressure compressors 13, 12 and the fan 11 via concentric driveshafts 19, 21, 22.

30 Figure 2 shows a bearing chamber 20 located at the rear end of a shaft 22 on which the low pressure turbine (17 in Figure 1) is mounted by means of a bearing 28. Two oil injectors 24, 26 provide oil to the bearing 28. The oil injector 24 provides a targetted jet of oil 30 to the inner

(surfaces of the bearing 28 whilst the oil injector 26 provides a film of oil to the outer surfaces of the bearing. The oil 30 from injector 24 is distributed around the bearing by oil scallops 40. Oil strainers or filters 5 42, 44 are provided to remove unwanted debris from the oil supply and a bearing compartment end cap 46 seals the rear of the bearing compartment 48. Oil exits the bearing compartment 48 through an oil scavenge pipe 50 fitted with a filter or strainer 44.

10 Oil is supplied to the injectors 24 and 26 through a common oil feed pipe 32. The supply of oil 30 is controlled by a control system (not shown) such that the amount of oil supplied is directly linked to the operating conditions of the engine, and in this embodiment to the requirements of 15 the bearing and its surroundings.

Now referring to Figure 3, an oil injector 54 is located in the bearing compartment 68 of the front bearing 66 of the low pressure turbine shaft 22. In this embodiment only one oil injector 54 is shown and this is fed with oil 20 through a feed pipe 56. The bearing 66 is located in the front bearing compartment 68 and is enclosed by a casing structure 70. Rotating seals 82 and 88 prevent oil from escaping from the bearing chamber 68.

An example of a lubrication system flow layout of 25 known type is shown in Figure 4. Oil is pumped from a tank 100 by a pump 102, driven by a motor 104. The motor is controlled by a control system (not shown) to maintain a constant pressure on the supply side of the lubrication system, as measured by pressure and temperature sensor 106. 30 It is envisaged that different supply pressures might be required for different operating conditions, and the control system would accommodate this. The pumped oil passes through an anti-syphon valve 108 and a filter 110. A magnetic chip detector 112 is located adjacent the filter

110 to trap ferrous debris. If the filter 110 becomes blocked, a spring-loaded bypass valve 114 opens so the oil can flow through a backup filter 116. A pressure drop sensor 118 across the filter 110 generates a warning signal
5 to indicate to an operator that the filter 110 is blocked.

A heat exchanger 120, in the form of a fuel-cooled oil cooler, is located in the main oil delivery line 122. This removes heat from the oil and uses it to increase the fuel temperature. The fuel inlet and outlet are shown by dashed
10 lines 124 and 126. Other types of heat exchanger could equally well be used in place of the fuel-cooled oil cooler, for example an air-oil heat exchanger.

The oil supply line 122 delivers oil to three components 130, 132, 134. It will be appreciated that other
15 lubrication systems may have fewer or more components. Associated with each of the components 130, 132, 134 is a variable flow restrictor, respectively 240, 242, 244, which controls the delivery of oil to each component 130, 132, 134. A control system (not shown) varies the opening of
20 each restrictor 240, 242, 244, thereby changing the pressure drop across it and controlling the flow rate of oil to each component 130, 132, 134 according to demand.

Surplus oil is removed from each component 130, 132, 134 through an associated variable restrictor 160, 162,
25 164. A pump 166, powered by a motor 168, is of similar type to that on the supply side, and maintains a substantially constant sink pressure to extract oil (and air) from each component 130, 132, 134. The system will be scavenging varying proportions of oil and air from the three
30 components 130, 132, 134 depending on the quantity of oil being supplied to each by the restrictors 240, 242, 244. The respective sensors 150, 152, 154 sense the pressure at the scavenge side of each component 130, 132, 134 (which will vary according to the proportions of oil and air in

(the fluid scavenged from each component) and the size of the aperture in each variable restrictor 160, 162, 164 is then adjusted accordingly to maintain effective scavenging. The oil scavenged by the pump 166 is returned to the tank
5 100.

A disadvantage of the lubrication system of Figure 4 is that as the aperture of the variable restrictor 240, 242, 244 changes the pressure drop across it also changes. This will upset the aim of the oil jet or spray so that the
10 oil is directed at a different part of the component 130, 132, 134. This means that optimum lubrication of each component 130, 132, 134 can only be achieved for one condition of its associated variable restrictor 240, 242, 244.

15 A further disadvantage of the lubrication system of Figure 4 is that the operation of a restrictor-based control system will be very complex. Restrictors relying on controlling pressure drop to vary volumetric or mass flow rate will be subject to non-linearities introduced by, for
20 example, variations in temperature (and hence viscosity) of the oil. Open loop control of such a system would require a very high fidelity model to predict the flow rates, which would have to take account of the upstream and downstream pressures and the temperature.

25 Figure 5 shows an embodiment of a lubrication system according to the invention, which seeks to overcome this disadvantage.

Many components shown in Figure 5 are identical to components in Figure 4, and are identified by the same
30 reference numbers. In particular, the scavenge side of the system operates identically to that described in the embodiment of Figure 4.

On the delivery side of the lubrication system oil is pumped from a tank 100 by a pump 102, driven by a motor

104. The motor is controlled by a control system (not shown) to maintain a constant pressure on the supply side of the lubrication system, as measured by pressure and temperature sensor 106. It is envisaged that different
5 supply pressures might be required for different operating conditions, and the control system would accommodate this. The pumped oil passes through an anti-syphon valve 108 and a filter 110. A magnetic chip detector 112 is located adjacent the filter 110 to trap ferrous debris. If the
10 filter 110 becomes blocked, a spring-loaded bypass valve 114 opens so the oil can flow through a backup filter 116. A pressure drop sensor 118 across the filter 110 generates a warning signal to indicate to an operator that the filter is blocked.

15 A heat exchanger 120, in the form of a fuel-cooled oil cooler, is located in the main oil delivery line 122. This removes heat from the oil and uses it to increase the fuel temperature. The fuel inlet and outlet are shown by dashed lines 124 and 126. Other types of heat exchanger could
20 equally well be used in place of the fuel-cooled oil cooler, for example an air-oil heat exchanger. An accumulator 128 is connected to the oil delivery line 122 to smooth out any fluctuations in the oil supply.

The oil supply line 122 delivers oil to three
25 components 130, 132, 134. It will be appreciated that other systems may have fewer or more components. Associated with each component 130, 132, 134 is an injector, respectively 140, 142, 144, which delivers oil in a jet or spray to the required part of its respective component 130, 132, 134.

30 The injectors 140, 142, 144 are variable aperture pulsative injectors of the type described in German patent application DE 100 01 055 A1. This type of injector can be switched by a control system (not shown) between an open state in which it will permit fluid flow and a closed state

in which it will prevent fluid flow. The size of the aperture in the open state can be separately controlled.

A control system (not shown) controls the size of the aperture of each injector 140, 142, 144 and the frequency
5 of its opening and closing so that oil is delivered to the components 130, 132, 134 in a series of discrete, time-based pulses. Pressure and temperature sensors 150, 152, 154 associated with each component 130, 132, 134 relay to the control system the temperature of the oil leaving the
10 component 130, 132, 134 and the control system responds to maintain the required oil flow to each component 130, 132, 134.

Figure 6 shows schematically a number of possible flow modulations for one of the injectors 140, 142, 144,
15 illustrating some of the possible variations in quantity and pattern of oil delivery. In each graph, the x-axis represents time and the y-axis represents the level of fluid flow.

(a), (b) and (c) show arrangements in which the
20 injector is in its open state for, respectively, one-quarter, one-half and three-quarters of the time and it will be apparent that the quantity of oil delivered to the injector's associated component in a given time period will be proportional to the area under the graph. The
25 arrangements of (a), (b) and (c) could also be combined, over time, in any desired way, as shown in (d). This type of flow modulation is commonly called "pulse width modulation".

(e) illustrates how the variable aperture of the
30 injector may be exploited. It will be apparent that this arrangement passes the same quantity of oil in a given time period as does (a), but by pulses of twice the duration, each delivering oil at half the flow rate.

Lastly, (f) illustrates the type of flow modulation

commonly called "variable mark/space". Relatively small, equally sized pulses of oil are delivered more frequently when more oil is required and less frequently when less oil is required.

5 The flow modulations illustrated in Figure 6 are not intended to be exhaustive. It will be recognised, by one skilled in the art, that the techniques of flow modulation described here may be varied and combined in any desired way, and may be supplemented or replaced by other similar
10 techniques not described here but which will be familiar to those skilled in the art.

 The type of injector 140, 142, 144 described above has a variable aperture. Other types of injector are known, and could be used in place of the variable aperture injector if
15 it was more suitable for a particular component 130, 132, 134. One such type of injector, commonly called a "multiple step" injector, has an aperture whose size can be adjusted in a number of discrete steps (typically three). Yet another type of injector has no adjustment for the aperture
20 size, and operates purely as a pulsative injector.

 A further feature, which may be incorporated into any of the types of injector 140, 142, 144 described above, allows a "bypass flow" of oil at all times, whether or not the main aperture of the injector 140, 142, 144 is open. In
25 this way a certain minimum flow of oil is maintained to the component 130, 132, 134 at all times, supplemented as required by the controlled, pulsed flow.

 A problem arises with the embodiment shown in Figure 5 if, for some reason, only air is being scavenged from one
30 of the components 130, 132, 134. If the restrictor does not close sufficiently or quickly enough, the resulting high volumetric flow rate into the pump 166 will effectively prevent any scavenging from the other components 130, 132, 134.

Figure 7 shows an alternative embodiment of a lubrication system according to the invention, which seeks to overcome this disadvantage.

Many components shown in Figure 7 are identical to 5 components in Figure 5, and are identified by the same reference numbers. In particular, the delivery side of the system operates identically to that described in the embodiment of Figure 5.

In the embodiment of Figure 7, oil is scavenged from 10 each component 130, 132, 134 by an associated scavenge pump 180, 182, 184 of the constant-volume type. Each scavenge pump 180, 182, 184 is driven by an associated motor 190, 192, 194. The speed of each motor 190, 192, 194 is controlled in response to the scavenge requirements of its 15 associated component 130, 132, 134 (as measured by sensors 150, 152, 154). The scavenged oil is returned to the tank 100. The scavenging from each component 130, 132, 134 is therefore matched closely to that component's requirements.

The scavenge pumps 180, 182, 184 may alternatively be 20 of the constant speed, variable volume type.

Claims

1. A lubrication system comprising
a reservoir for lubricant;
5 lubricant delivery means to deliver lubricant to at
least one component to be lubricated;
an injector associated with each component;
a control system to control the opening and closing of
each injector;
10 scavenge means to remove surplus lubricant from around
each component and return it to the reservoir;
in which each injector comprises a pulsative injector,
controlled by the control system to deliver lubricant to
its associated component in a series of discrete pulses.
- 15 2. A lubrication system as claimed in Claim 1, in which
at least one injector is a multiple step injector.
3. A lubrication system as claimed in Claim 1 or Claim 2,
in which at least one injector is a variable aperture
injector.
- 20 4. A lubrication system as claimed in any preceding
claim, in which at least one injector has a minimum flow
bypass.
5. A lubrication system as claimed in any preceding
claim, in which the control system controls the supply of
25 lubricant to at least one component in response to the
operating conditions of that component.
6. A lubrication system as claimed in any preceding
claim, in which the lubricant delivery means comprises a
pump.
- 30 7. A lubrication system as claimed in Claim 6, in which
the lubricant delivery means comprises at least one
accumulator to smooth the fluctuations in pressure arising
out of the pulsative delivery of lubricant.

8. A lubrication system as claimed in any preceding claim, in which the scavenge means comprises a variable speed scavenge pump maintaining a substantially constant pressure head and a restrictor associated with each component.
9. A lubrication system as claimed in Claim 8, in which at least one restrictor is a variable restrictor controlled by the control system in response to the operating conditions of its associated component.
10. A lubrication system as claimed in any of Claims 1 to 7, in which the scavenge means comprises a pump associated with each component, each pump being controlled by the control system in response to the operating conditions of its associated component.
11. A lubrication system as claimed in Claim 10, in which at least one pump is a constant volume pump.
12. A lubrication system as claimed in Claim 10 or Claim 11, in which at least one pump is a constant speed, variable volume pump.
13. A gas turbine engine including a lubrication system as claimed in any preceding claim.
14. A lubrication system substantially as described in this specification, with reference to and as shown in Figures 2, 3, 5 and 7 of the accompanying drawings.
15. A method for lubricating a component of a gas turbine engine, comprising the steps of
- storing lubricant in a reservoir;
 - delivering the lubricant to a pulsative injector associated with the component;
 - controlling the pulsative injector to deliver lubricant to the component in a series of discrete pulses;
 - scavenging surplus lubricant from the component and returning it to the reservoir.

16. A method for lubricating a component of a gas turbine engine, comprising the steps of

storing lubricant in a reservoir;

delivering the lubricant to a pulsative injector
5 associated with the component;

sensing at least one first condition of the component;

controlling the pulsative injector, in response to at least one first condition of the component, to deliver lubricant to the component in a series of discrete pulses;

10 scavenging surplus lubricant from the component and returning it to the reservoir.

17. A method according to Claim 15 or Claim 16 in which the scavenging of surplus lubricant comprises the steps of

sensing at least one second condition of the
15 component;

controlling a scavenge pump associated with the component, in response to at least one second condition of the component, to scavenge surplus lubricant from the component and return it to the reservoir.

20 18. A method according to Claim 17 in which the second condition is one or more selected from the group comprising a pressure, a temperature, a heat flux, an engine speed.

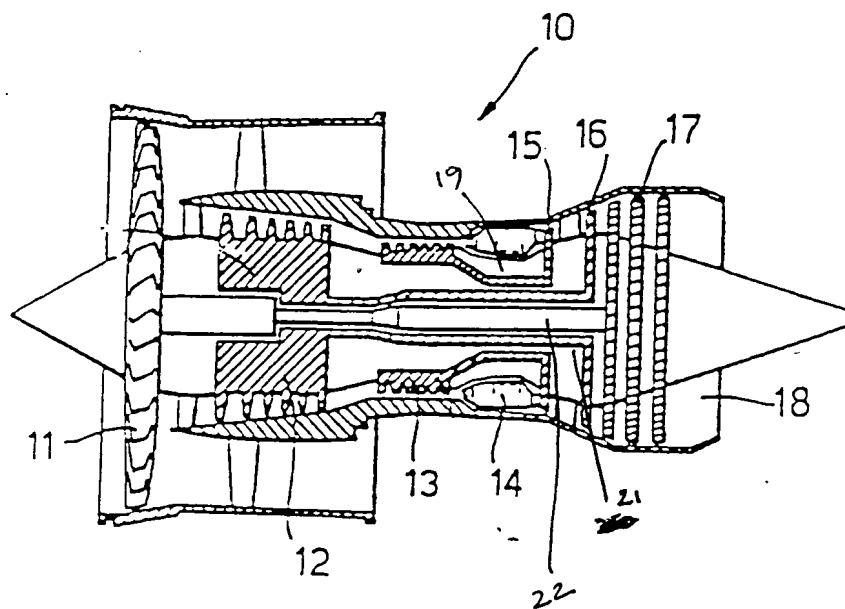
19. A method of lubricating a component of a gas turbine engine in which the component is lubricated by a
25 lubricating system according to any of claims 1 to 14.

Abstract

(refers to Figure 7)

A lubrication system comprises a reservoir for
5 lubricant, lubricant delivery means to deliver lubricant to
at least one component to be lubricated, an injector
associated with each component, a control system to control
the opening and closing of each injector and scavenge means
to remove surplus lubricant from around each component and
10 return it to the reservoir. Each injector comprises a
pulsative injector, controlled by the control system to
deliver lubricant to its associated component in a series
of discrete pulses. In a preferred embodiment both the
delivery of oil to each component and the scavenging of oil
15 from each component are controlled in response to sensed
conditions of that component.

FIGURE 1



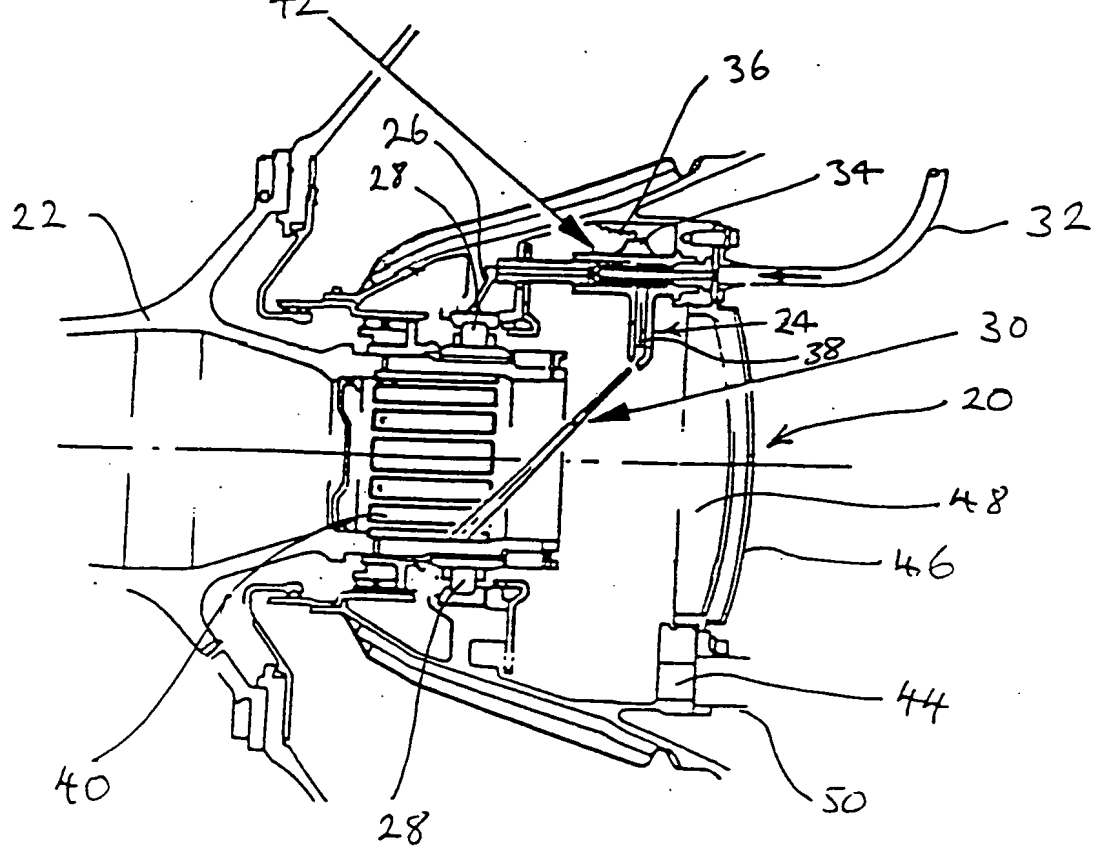
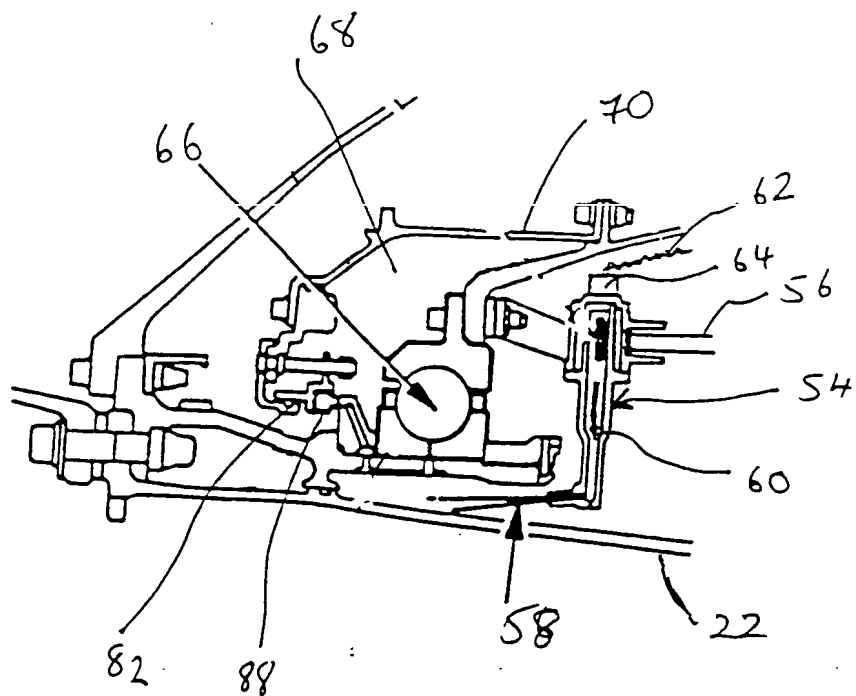


FIGURE 3



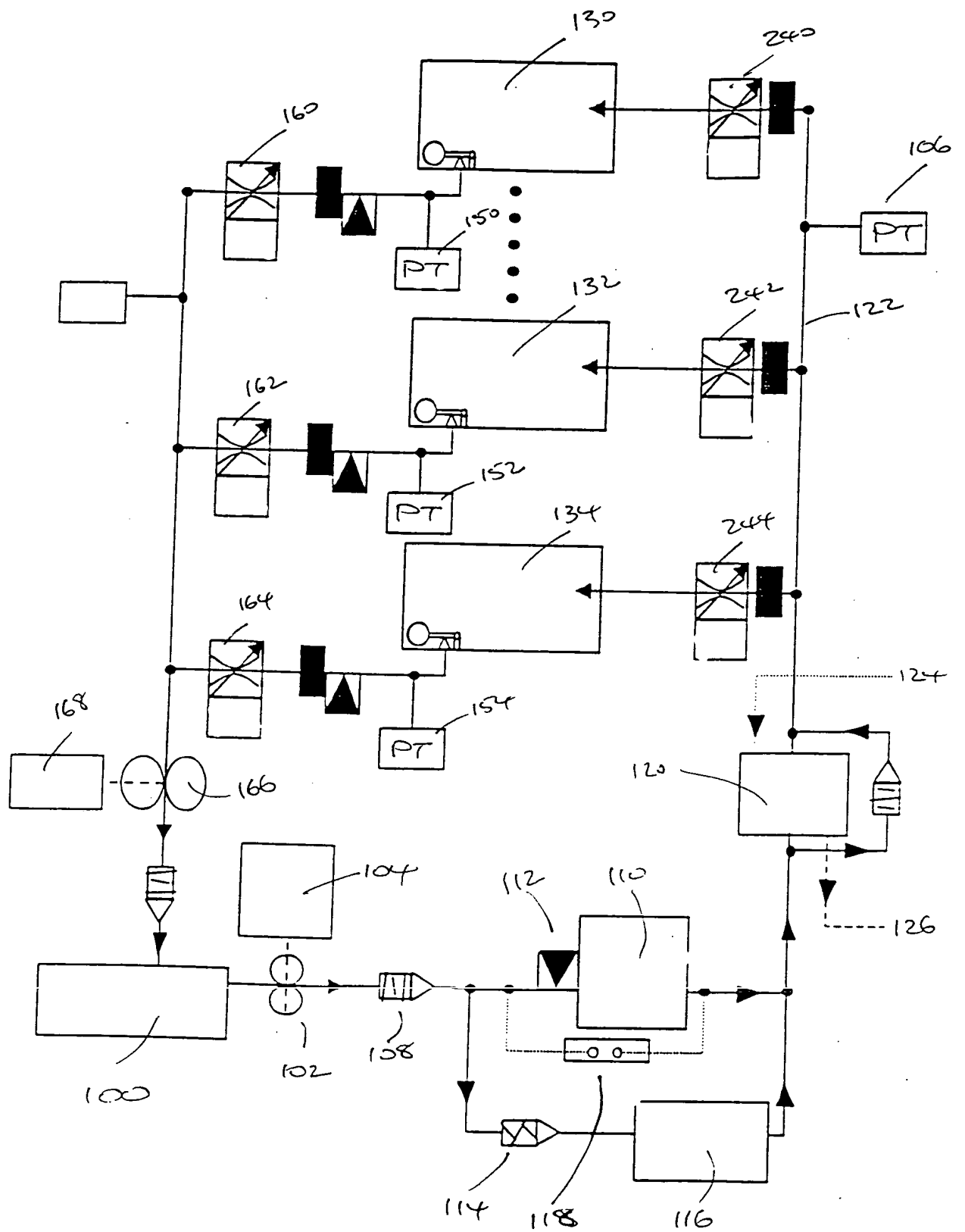


Figure 4

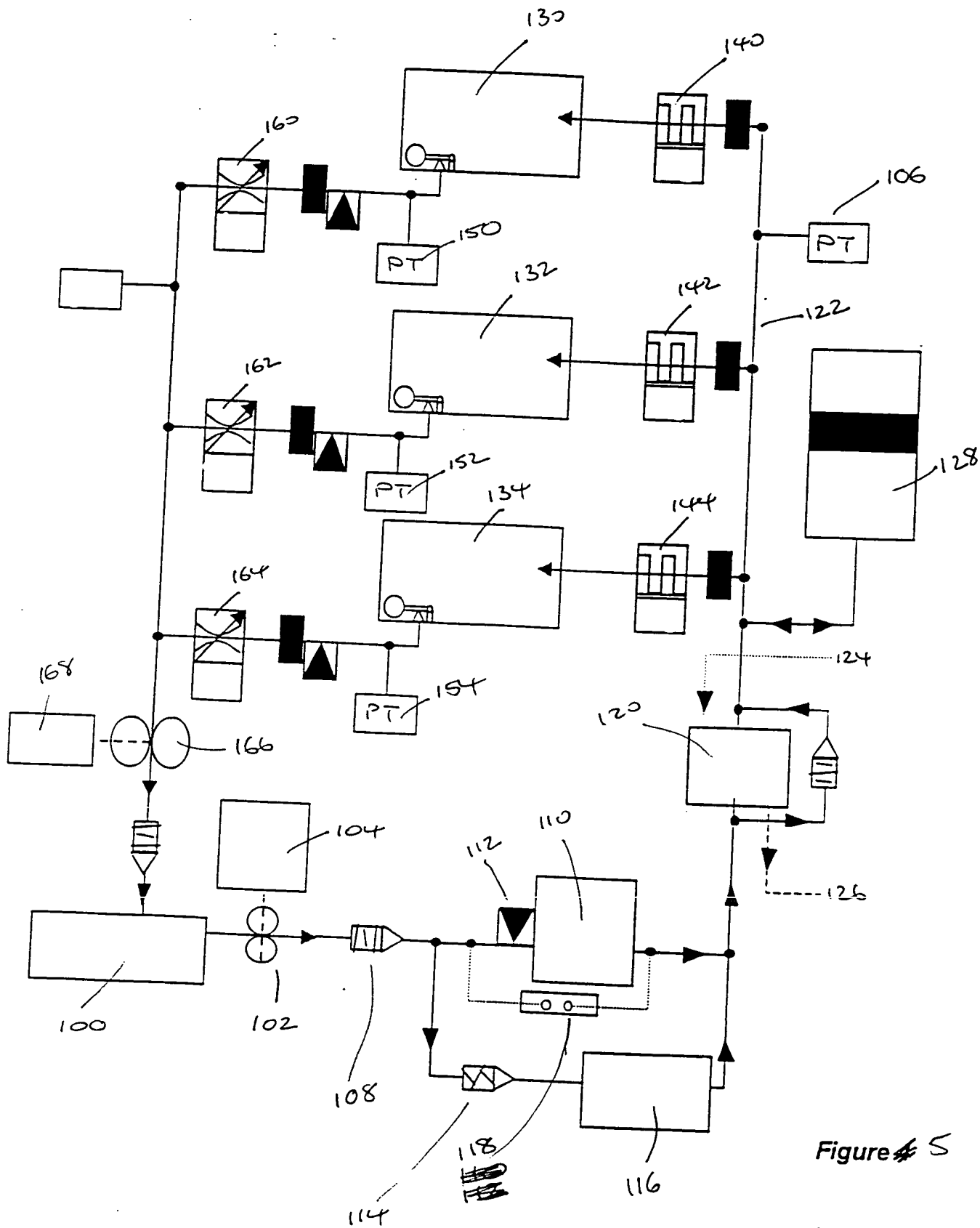
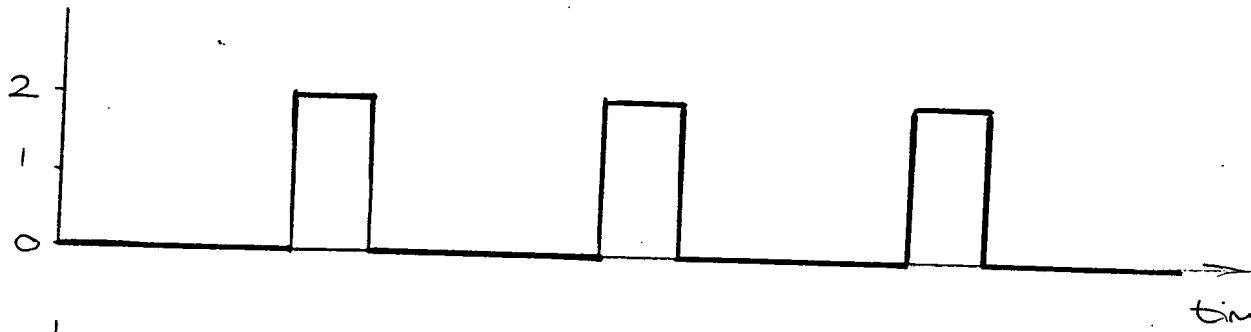


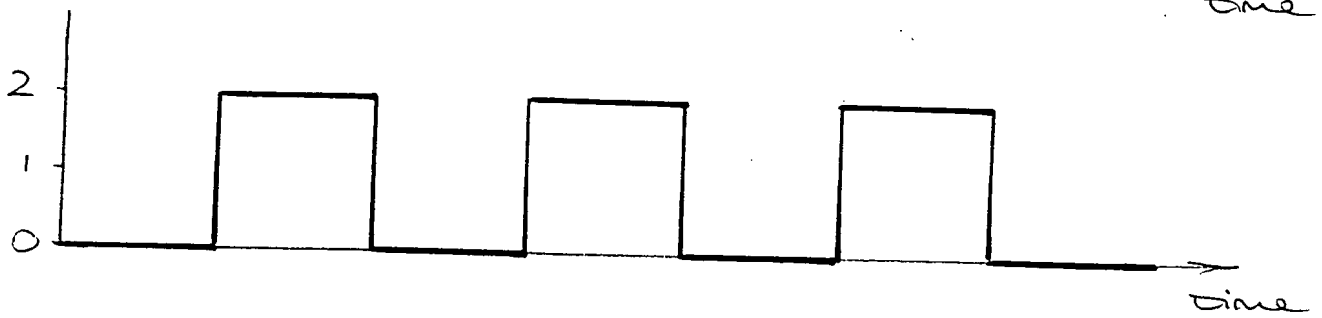
Figure # 5

Figure 6

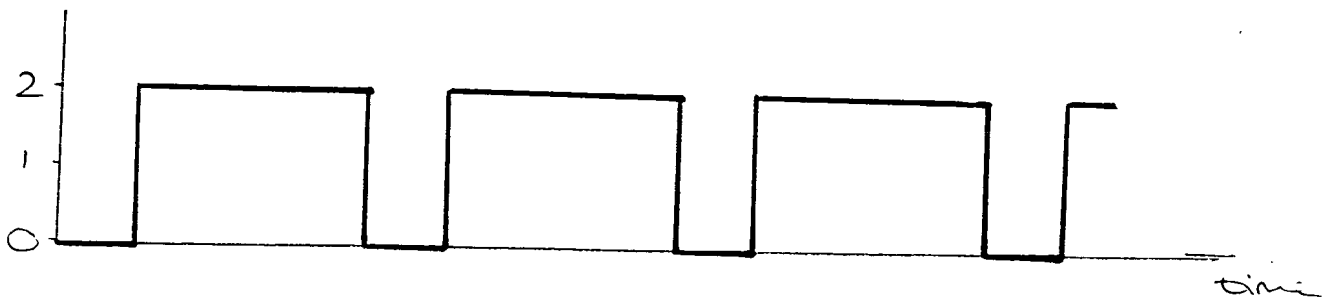
(a)



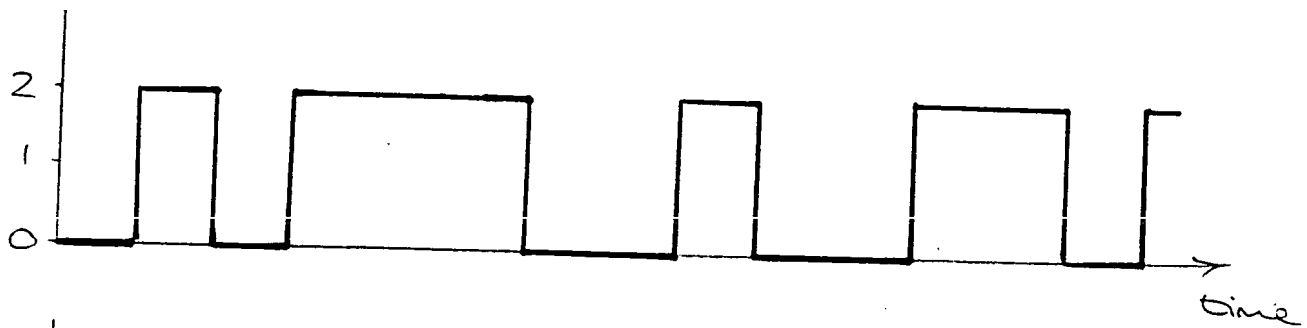
(b)



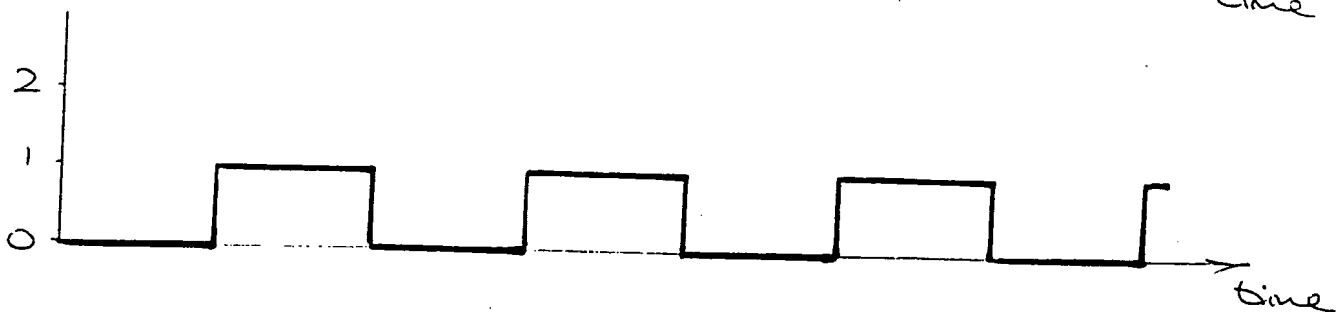
(c)



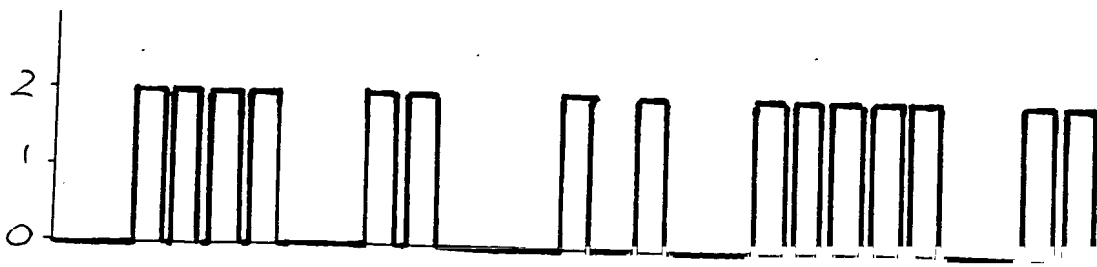
(d)



(e)



(f)



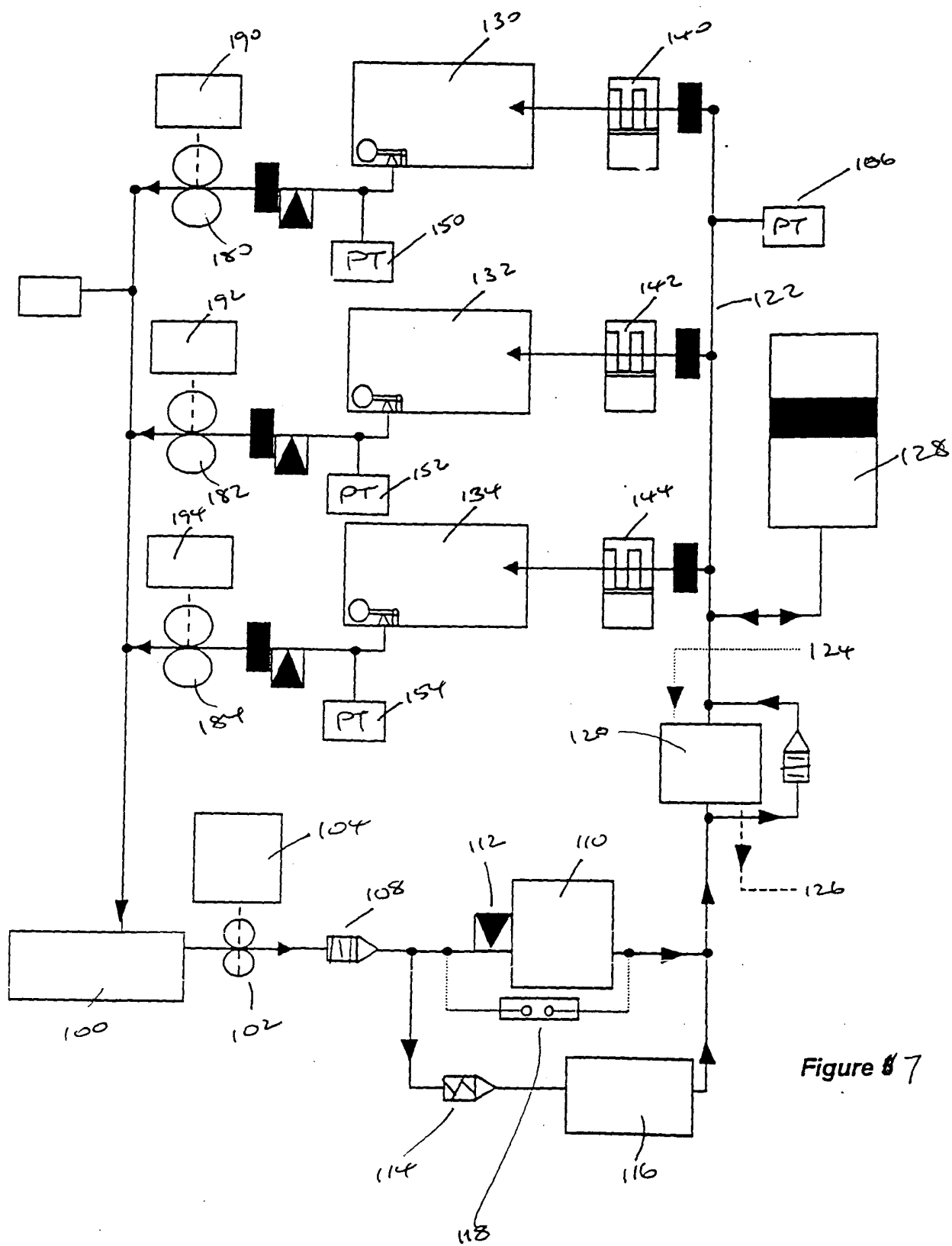


Figure # 7

